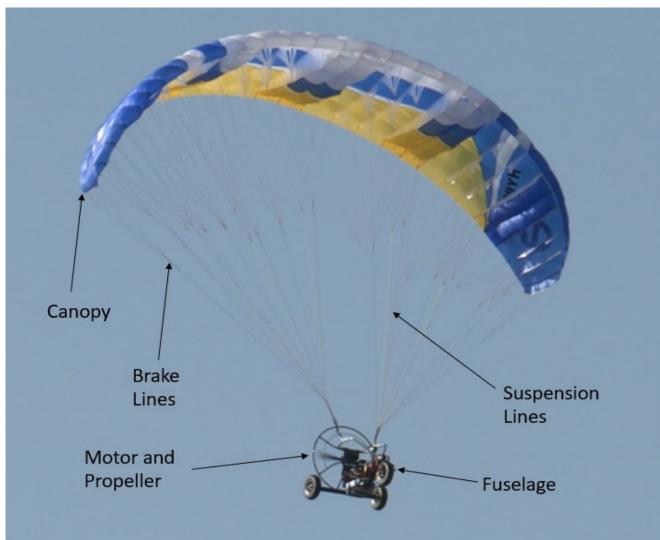
Aero-propulsive characterization of Paramotor sUAS

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Introduction

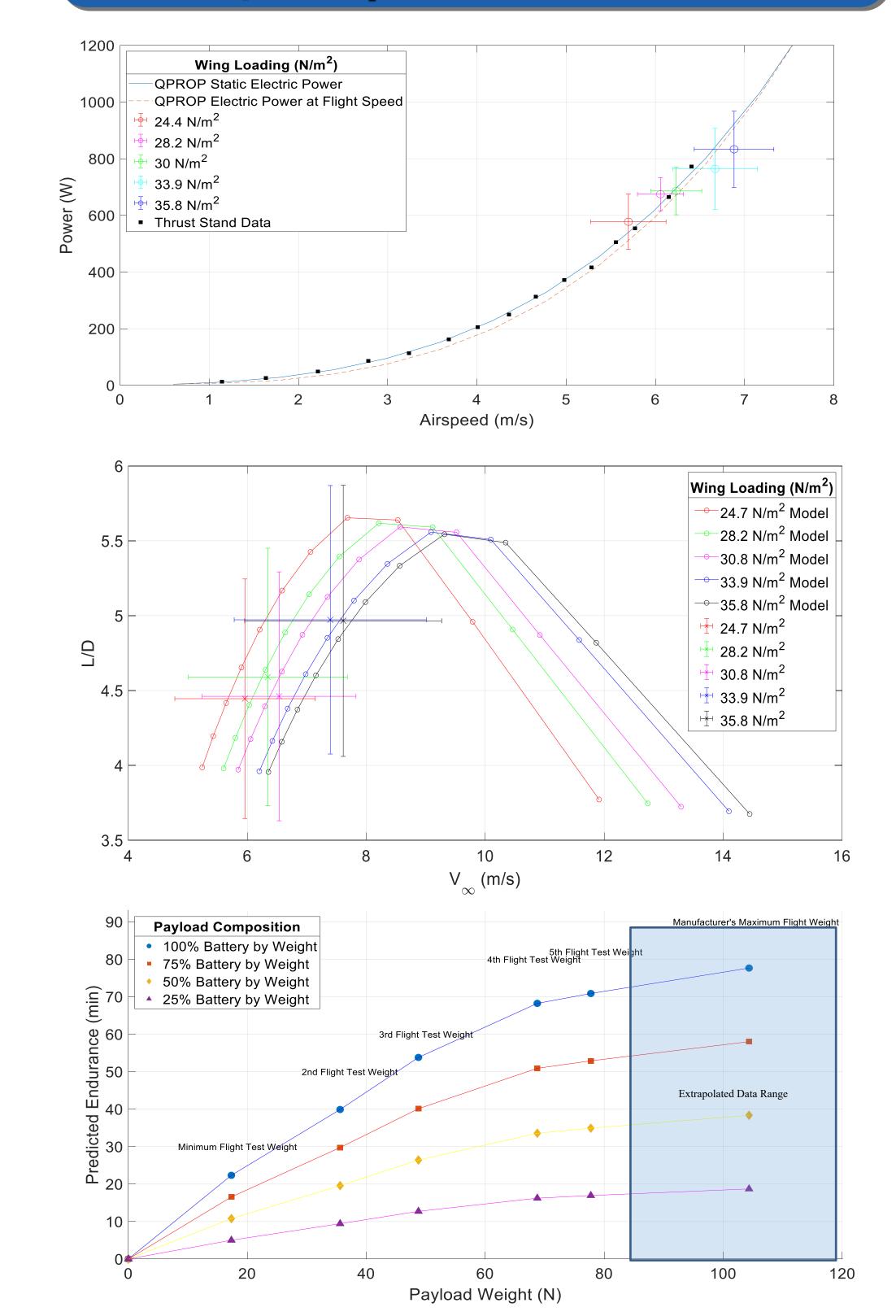
This poster summarizes the work documented by the authors in Ref 1.



Paramotor Characteristics:

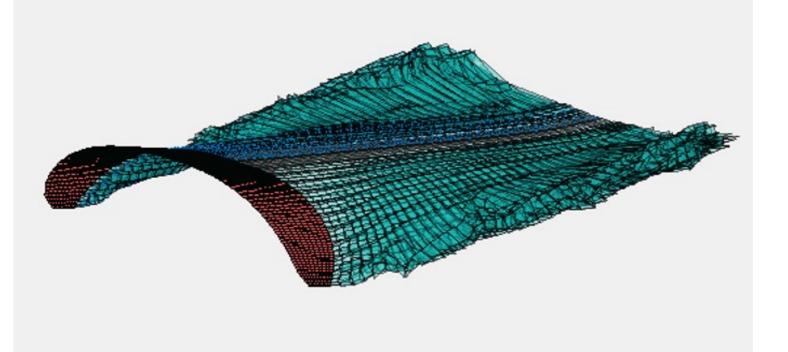
- Parafoil canopy
- Fuselage with propeller thrust system
- Brake lines for yaw control
- Single speed (function of wingloading) [2]
- Pendulum stability
- High payload capacity
- Short takeoff [5]

Results / Analysis



Motivation:

 Little flight test data exists in



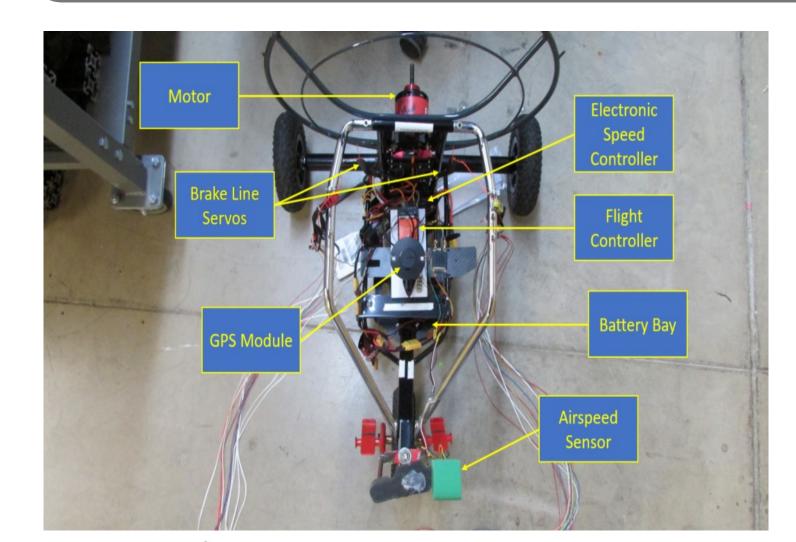
- Iterature
 Apply models as a design space aid
- Capture aerodynamic behavior for predictions

Objectives

Research addressed four primary questions about sAUS Paramotors:

- 1. What unique challenges are there when flight testing paramotor sUAS?
- 2. How accurate are traditional fixed wing modelling techniques at capturing paramotor behavior and how can they be adapted to understand vehicle dynamics?
- 3. How does paramotor performance change with wing loading? What does this mean for potential applications of paramotor sUAS?
- 4. What is the mission space for this paramotor in terms of endurance and payload and how can we use models to find this space for other paramotors?

Methodology



Flight Testing:

- Identify and adapt to flight test challenges
- Fly in glide, climb, and SLF at several weights

Future Objectives

- Additional flight test data is required to confirm trends and reduce uncertainty particularly in the climb rate investigation
- An accurate model for climb rate and testing the empirical 54% interference drag assumption would complete the



- Record vehicle
 behavior and derive
 trends
- aerodynamic model
- Building a new paramotor and testing endurance predictions of the method described would confirm the utility of this work as a design space aid

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References

Haefner, Alex C. Characterization of the Aero-Propulsive Performance of a sUAS Paramotor, AIAA, 24 June 2024.
 J. Chambers, "Longitudinal dynamic modeling and control of powered parachute aircraft," Rochester Institute of Technology, 2007.
 N. Slegers, E. Beyer, and E. Costello, "Use of Variable Incidence Angle for Glide Slope Control of Autonomous Parafoils," Journal of Guidance, Control, and Dynamics, 2008.

[4] O. Yakimenko, "Precision Aerial Delivery Systems- Modeling, Dynamics, and Control," American Institute of Aeronautics and Astronautics, 2015 [5] S. Jovicic, S. Tirnanic, S. Ilic, and N. Brkljac, "Applications of powered paragliders in military, police special forces, searching and rescue units," 6th International Scientific Conference on Defensive Technologies, 2014.



